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Averge tooth load

Force of on the planet carrier divided over 4 planet wheels. Assuming that all the forces are devided equal over each planet wheel $F_{gearring} \coloneqq \frac{M_{planet} \cdot \frac{1}{n_1}}{(R_{gearring})}$ $F_{gearring} = 1041.667 \ kN$

The avarge radial load on the tooth of the planet wheel $F_{planet} := \frac{M_{planet} \cdot \frac{1}{n_1}}{(R_{gearring} - R_{planetw})}$ $F_{planet} = 2083.333 \text{ kN}$

Radial load on the planet carrier $F_{rad} = 290 \ \textbf{kN}$

Even vragen aan waar deze vandaan komen

Add. load on planet 1 $F_{1add} = 1$	22 <i>kN</i>	Add. load on planet	3 $F_{3add} = -134 \ kN$
Add. load on planet 2 $F_{2add} = -$	-40 kN	Add. load on planet	$4 \qquad F_{4add} \coloneqq 53 \ \mathbf{kN}$
Force with added load 1	F_{1py}	$:= F_{1add} + F_{planet}$	$F_{1py} = (2.205 \cdot 10^3) \ kN$
Force with added load 2	F_{2py}	$:= F_{2add} + F_{planet}$	$F_{2py} = (2.043 \cdot 10^3) \ kN$
Force with added load 3	F_{3py}	$:= F_{3add} + F_{planet}$	$F_{3py} = (1.949 \cdot 10^3) \ kN$
Force with added load 4	${F}_{4py}$	$:= F_{4add} + F_{planet}$	$F_{4py} = \left(2.136 \cdot 10^3\right) kN$
		$+F_{2py}+F_{3py}+F_{4py}\Big)$	$F_{avx} = 2083.583 \ kN$
Force with added load 1 in %	${F}_{1p}$:	$=\frac{F_{planet}}{F_{1py}} \qquad F_{1p}=$	94.468%
Force with added load 2 in %		F_{2py}	101.958%
Force with added load 3 in %	F_{3p} :	$= \frac{F_{planet}}{F_{3py}} \qquad \qquad F_{3p} =$	106.874%
Force with added load 4 in %	F_{4p} :	$=\frac{F_{planet}}{F_{4py}} \qquad F_{4p}=$	97.519%

The three equations

Moment balance around the axis of the planetary carrier:

The moment balance around the axis of the planetary carrier tells us that the sum of all moments around this axis must be equal to the total moment on the planetary carrier. So:

 $M_pI = (F_1 + F_2 + F_3 + F_4) \times (2/3)$

Here:

 $M_pl = 2500 \text{ kNm}$, the moment on the planetary carrier.

D = 1.2 m, the diameter of the gear ring.

F_1, F_2, F_3, F_4 are the unknown forces on the teeth of the gear system.

Force balance in the x-direction:

 $F_1 - F_2 \times tan(a) - F_3 + F_4 \times tan(a) = Frad$

Here, Frad = 290 kN is the radial load on the planetary carrier. See the freebody diamgram

Force balance in the y-direction:

 $-F_1 \times tan(a) - F_2 + F_3 \times tan(a) + F_4 = 0$

Setup first equations of motion

We now have three equations: the moment balance and force balances in the x and y directions. To solve this system, we need four equations. However, the moment balance provides only one additional equation.

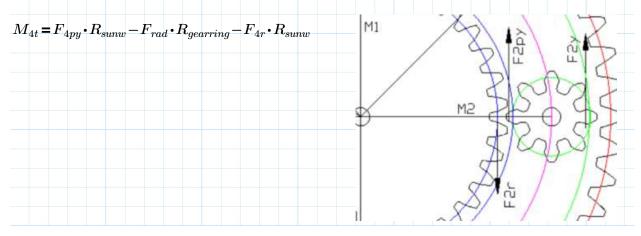
To obtain the fourth equation, we can leverage the geometric symmetry of the gear system. We can assume that the forces are evenly distributed across the planetary gears, and that the radial load is equal on each tooth. With this geometric assumption, we can derive a fourth equation.

Step 1: Adding extra geometric symmetry for uniform force distribution Consideration of radial distances: We can consider the radial distances between the axis of each gear and the axis of the planet carrier. These radial distances can help us establish additional equations based on geometric ratios. The geometric relationships of the system, particularly the radial distances between the axes of the gears and the planet carrier. These radial distances can assist us in establishing additional equations.
Assuming that:
 The radial distance between the axis of each gear and the axis of the planet carrier is known. that the arm are equal long and the tolerances and geometric shape of the wheels are the same.
With this information, we can formulate 4 moment equations based on the radial forces acting on the gears and the moment arms between the axes of the gears and the planet carrier.
The moment equations might look as follows:
Moment balance around the planetary carrier arm 1:
$M_{1t} = F_{1py} \cdot R_{sunw} - F_{rad} \cdot R_{gearring} - F_{1r} \cdot R_{sunw}$
Moment balance around the planetary carrier arm 2:
$M_{2t} = F_{2py} \cdot R_{sunw} - F_{rad} \cdot R_{gearring} - F_{2r} \cdot R_{sunw}$

Moment balance around the planetary carrier arm 3:

$$M_{3t} = F_{3py} \boldsymbol{\cdot} R_{sunw} - F_{rad} \boldsymbol{\cdot} R_{gearring} - F_{3r} \boldsymbol{\cdot} R_{sunw}$$

Moment balance around the planetary carrier arm 4:



Moment balance around the planetary carrier arm 4:

equations:	ient equa		maunx	101111.			
	$M1 = F_{12}$	$_{py}ullet R_{sunw}$ -	$-F_{rad} \cdot I$	$\mathbf{R}_{gearrin}$	$g - F_{1r} \cdot R_{st}$	inw	
	$M2 = F_{2}$	$_{py}ullet R_{sunw}$ -	$-F_{rad} \cdot I$	$\mathbf{R}_{gearrin}$	$_{g}-F_{2r}ullet R_{sv}$	inw	
	$M3 = F_{33}$	$_{py}ullet R_{sunw}$ -	$-F_{rad} \cdot I$	$\mathbf{R}_{gearrin}$	$_{g}-F_{3r}ullet R_{st}$	ınw	
	$M4 = F_{4j}$	$_{py}oldsymbol{\cdot}R_{sunw}$ -	$-F_{rad} \cdot I$	$R_{gearrin}$	$_{g}-F_{4r}ullet R_{st}$	ınw	
coefficient matrix		R_{sunw}	0	0	0]		
coefficient matrix	(Δ. Δ.	0	R_{sunw}	0	0		
		0	0 1	R_{sunw}	0		
		0	0	0	R _{sunw}		
		$\int -F$	$_{1py} \cdot R_{su}$	$_{inw}+F_{i}$	$_{ad} \cdot R_{gearrin}$		
vector matrix		$b_1 := -F$	$r_{2py} \cdot R_{su}$	$u_{nw} + F_r$	$r_{ad} \cdot R_{gearrin}$	ng	
		-F	$_{3py} \cdot R_{su}$	$w_{nw} + F_{\eta}$	$r_{ad} \cdot R_{gearrin}$ $r_{ad} \cdot R_{gearrin}$	ng	
		$\lfloor -F \rfloor$	$_{4py}$ • R_{su}	$_{nw}+F_{r}$	$r_{ad} \cdot R_{gearrin}$		
						[1045 103	31
			1 .			$-1.045 \cdot 10$ $-883 \cdot 333$	
Radiaal krachten		F_{radt} =	$=A^{-1} \cdot b$	1	F_{rad}	$t = \begin{vmatrix} 0.00.000 \\ -789.333 \end{vmatrix}$	kN
						$t = \begin{bmatrix} -1.045 \cdot 10^{5} \\ -883.333 \\ -789.333 \\ -976.333 \end{bmatrix}$]
						${}_{h} = \begin{bmatrix} -2.872 \cdot 10 \\ -2.427 \cdot 10 \\ -2.169 \cdot 10 \\ -2.682 \cdot 10 \end{bmatrix}$	3
Horizontaal krach	nten	<i>F</i> ::	= <i>F…</i> •	1	— <i>F</i>	$_{l} = -2.427 \cdot 10$	3 kN
		- raan	- raai	$\tan{(\alpha}$	$_{1}$ $^{-raa}$	n -2.169 • 10	3
						$[-2.682 \cdot 10]$	3
				1		$-3.056 \cdot 10^{\circ}$	
Pressure angle for	orce	$P_{force} \coloneqq F$	radh•		$P_{force} =$	$\begin{bmatrix} -3.056 \cdot 10^3 \\ -2.583 \cdot 10^3 \\ -2.308 \cdot 10^3 \\ -2.855 \cdot 10^3 \end{bmatrix}$	kN
			cc	$\operatorname{s}(\alpha_1)$		$-2.308 \cdot 10^{-2}$	
						[-2.000.10]	

Convert the moment equations into matrix form. We have four moment equations:

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